

Gas turbine intake air hybrid cooling systems and their rational designing

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Abstract. The general trend to improve the fuel efficiency of gas turbines (GT) at increased ambient temperatures is turbine intake air cooling (TIAC) by exhaust heat recovery chillers. The high efficiency absorption lithium-bromide chillers (ACh) of a simple cycle are the most widely used, but they are not able to cool intake air lower than 15°C because of a chilled water temperature of about 7°C. A two-stage hybrid absorption-ejector chillers (AECh) were developed with ejector chiller as a low temperature stage to provide deep air cooling to 10°C and lower. A novel trend in TIAC by two-stage air cooling in chillers of hybrid type has been proposed to provide about 50% higher annual fuel saving in temperate climatic conditions as compared with ACh cooling. The advanced methodology to design and rational distribute the cooling capacity of TIAC systems that provides a closed to maximum annual fuel reduction without oversizing was developed.

1 Introduction

The efficiency of gas turbines (GT) decreases with arising the ambient air temperature at their inlet [1, 2]. The general trend to improve the fuel efficiency of GT at increased ambient air temperatures is turbine intake air cooling (TIAC) by exhaust heat recovery chillers [3]. A reduction of the chiller sizes with maximum annual fuel saving is possible due to rational design cooling capacity excluding oversizing [4, 5] and its rational distribution with small deviation of current loads from a design value [6, 7]. In order to realize this the overall band of current cooling loads is to be divided into two ranges: the first unstable load range, following the fluctuations of current loads (thermal "turbulences"), and the second range of comparatively stable loads ("laminarized" thermal load range). The cooling capacity of the chillers is to be designed to cover the thermal "turbulences" by the absorption lithium-bromide chiller (ACh) chillers with a high coefficient of performance (COP) not effected by load fluctuations considerably [8]. The further air subcooling takes place within the comparatively stable "laminarized" thermal load range) and can be covered by ejector chiller (ECh) as the most simple in design and cheapest but considerably effected by load changes [9]. The application of such hybrid absorption-ejector chiller (AECh) enables to cover actual loading in two-stage air cooler with boost high temperature water stage and low temperature refrigerant stage [10].

The purpose of the study is to develop the advanced hybrid TIAC systems and the improved methodology of

their designing with rational distribution of the overall design cooling capacity between unstable ("turbulent") thermal load range for ambient air precooling in the boost high temperature stage of the air cooler (AC) by ACh and a stable ("laminarized") load range for further air subcooling to the target temperature in the low temperature stage by ECh that provides practically twice reduction of a design boost thermal load and about 50% higher annual fuel saving as compared with ACh gained due to applications of TIAC systems with hybrid AECh.

2 Literature review

In a number of investigations the combustion engine intake air cooling (EIAC) including TIAC is considered as subtechnologies for combined cooling, heating and power (CCHP) [11], or trigeneration [12, 13]. A lot of researches are focused to improve the performance of air cooling systems by intensification of heat transfer in evaporators and condensers [14].

The technical innovations in waste heat recovery [15, 16] including transport application [17, 18] might be successfully applied in TIAC: two-stage intake air cooling [10], deep exhaust heat utilization [19, 20]. The heat potential for converting in refrigeration can be increased due to low-temperature condensation [21, 22].

Advanced methods as ANSIS [23] and statistical methods for processing monitoring data can be used for optimizing the cooling loads according site climatic conditions [24, 25] and along ship voyages [26]. The sinusoidal curve was proposed for daily thermal load fluctuations [27] to match current cooling demands.

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Practically all the typical design methods [28, 29] issue from the assumption of a design cooling capacity to cover maximum cooling needs over the full range of yearly operating conditions, that inevitable leads to considerable cooling system oversizing and requires to define a correct design cooling load excluding overestimation.

3 Research methodology

A reduction of the chillers design cooling capacity is possible by determining its rational value to provide closed to maximum annual fuel saving as the first step of the methodology for designing the TIAC system and further distribution of the available cooling capacity in response to the current demands as the second step.

The annual fuel saving ΣB_e of the GT due to inlet air cooling is assumed as a criterion to determine a rational design cooling capacity Q_0 of the TIAC system. With this the current fuel reduction B_e have been summarized over the year:

$$\Sigma B_e = \Sigma(\Delta t_a \cdot \tau) \cdot b_{ei} \cdot N_e \cdot 10^{-3}, \quad (1)$$

where: $\Delta t_a = t_{amb} - t_{a2}$ – current intake air temperature drop, K or °C; t_{amb} and t_{a2} – ambient air and air temperature at the air cooler outlet, K or °C; N_e – turbine power output, kW; τ – time interval, h; b_{ei} – specific fuel reduction for 1K (1°C) air temperature drop, assumed 0.7 g/(kWh·K) for UGT10000 [30].

It is preferably to analyze the fuel reduction in dependence on specific cooling capacity q_0 as the overall its value Q_0 referred to air mass flow rate $G_a = 1 \text{ kg/s}$:

$$q_0 = Q_0 / G_a \text{ or } q_0 = \xi \cdot c_{ma} \text{ kW}/(\text{kg/s}) \text{ or kJ/kg}, \quad (2)$$

where: ξ – specific heat ratio; c_{ma} – moist air specific heat, kJ/(kg·K).

According to the method developed the fluctuations of the current effect in GT fuel reduction B_e are considered by the rate of their annual increment ΣB_e as relative annual fuel saving increment $\Sigma B_e / Q_0$ referred the cooling capacity needed. A such methodological approach makes it possible to increase the accuracy of the results due to excluding the approximation of the current changeable values of B_e . This is a principally novelty versus a generally accepted approach to cover the maximum current demands to reach the maximum annual value ΣB_e that leads to oversizing.

There are two methods developed: the first – by using the annual fuel reduction ΣB_e dependence on the design cooling capacity of the chiller to choose its rational value $Q_{0, \text{rat}}$, that provides closed to maximum annual fuel reduction ΣB_e , and the second – according to the maximum rate of annual fuel reduction ΣB_e increment $\Sigma B_e / Q_0$ to choose optimum design cooling capacity $Q_{0, \text{opt}}$, that provides minimum sizes of the chiller and TIAC system (Fig.1,a).

The rational value of design cooling capacity $Q_{0, \text{rat}}$, providing a closed to maximum annual fuel reduction ΣB_e is associated with the second maximum rate of annual fuel reduction ΣB_e increment within its range

beyond the first maximum rate: $Q_0 > Q_{0, \text{opt}}$ and $\Sigma B_e > \Sigma B_{e, \text{opt}}$ accordingly. With this a relative parameter $\Sigma(B_e - B_{e, \text{opt}}) / Q_0$ is used as indicator to choose a rational value $Q_{0, \text{rat}}$ (Fig.1,b).

The optimum $Q_{0, \text{opt}}$ and rational $Q_{0, \text{rat}}$ cooling capacities for $t_{a2} = 10, 15$ and 20°C were calculated for temperate climatic conditions of Voznesensk, Nikolaev region, southern Ukraine, 2017 year (Fig. 1).

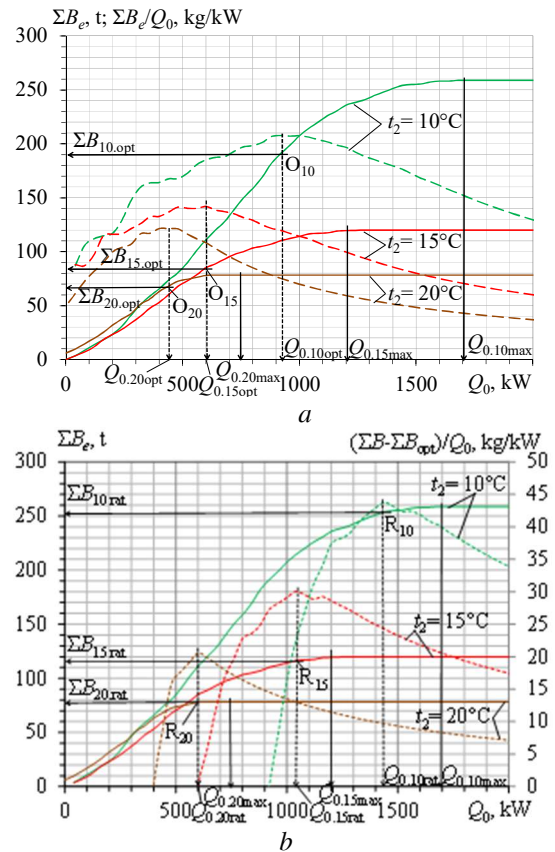


Fig.1. Annual fuel reduction ΣB_e and its relative values $\Sigma B_e / Q_0$ referred to design cooling capacity Q_0 over the whole range of ΣB_e (a) and values $\Sigma(B_e - B_{e, \text{opt}}) / Q_0$ beyond the optimum values of $\Sigma B_{e, \text{opt}}$ and $Q_{0, \text{opt}}$ (b) for cooling ambient air to $t_{a2} = 10, 15$ and 20°C .

A maximum rate of annual fuel reduction ΣB_e increment $\Sigma B_e / Q_0$ for $t_{a2} = 10^\circ\text{C}$ takes place at the optimum design cooling capacity $Q_{0, \text{opt}}$ of about 900 kW (Fig.1, a). A maximum rate of annual fuel reduction increment $\Sigma(B_e - B_{e, \text{opt}}) / Q_0$ within the range beyond the value $\Sigma B_{e, \text{opt}} = 190 \text{ t}$ corresponding to $Q_{0, \text{opt}} = 900 \text{ kW}$ takes place at the rational design cooling capacity $Q_{0, \text{rat}} = 1450 \text{ kW}$ and provides annual fuel reduction $\Sigma B_{e, \text{rat}} = 250 \text{ t}$ that is very closed to its maximum value 260 t but at a reduced design cooling capacity $Q_{0, \text{rat}} = 1450 \text{ kW}$ less than $Q_{0, \text{max}} = 1700 \text{ kW}$ by 15 %.

The rational distribution of a design cooling capacity in response to the current thermal loads, as the second step of the methodology, requires comparing the available cooling capacity of the chillers with current cooling loads to determine the excessive available cooling capacity, revealed at the lowered current thermal loads on the air cooler (AC) at the inlet of GT, to cover the peaked current thermal loads.

Because of great uncertainty of unstable boost ("turbulent") load range magnitude its design value

$q_{0,b}$ should be determined by a remaining principle as a difference between the overall design cooling capacity $q_{0,10}$ for the whole process of cooling the ambient air to the target temperature $t_{a2} = 10\text{ }^{\circ}\text{C}$ and its basic stable ("laminar") load range $q_{0,10-15}$ for subcooling air from a threshold air temperature of about $15\text{ }^{\circ}\text{C}$ after ACh to $t_{a2} = 10\text{ }^{\circ}\text{C}$: $q_{0,b} = q_{0,10} - q_{0,10-15}$, where $q_{0,10-15} = q_{0,10} - q_{0,15}$.

4 Results

The further development of the methodology of TIAC system designing involves distribution of the overall design cooling capacity between unstable ("turbulent") thermal load range for ambient air precooling in the boost high temperature stage of the air cooler (AC) by ACh and a stable ("laminarized") load range for further air subcooling to the target temperature in the low temperature stage by ECh (Fig. 2).

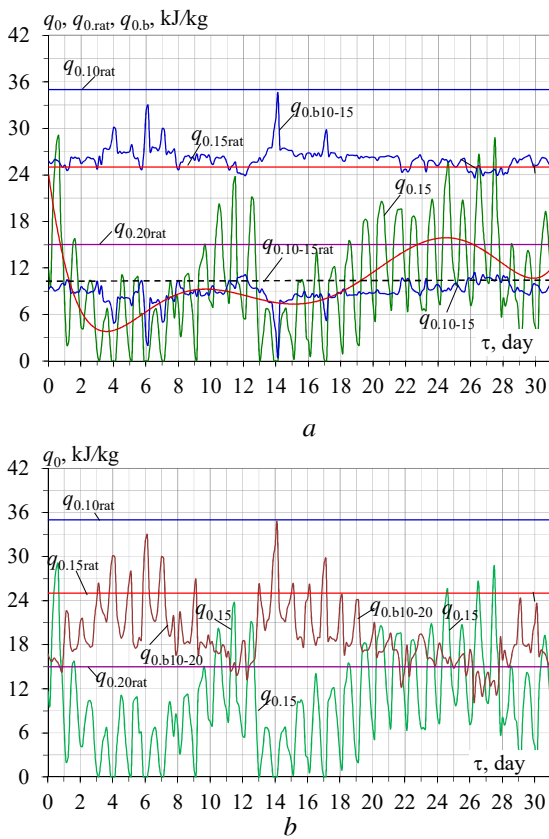


Fig.2. Current values of specific cooling capacities $q_{0,15}$ needed for cooling ambient air to $t_{a2} = 15\text{ }^{\circ}\text{C}$, rational cooling capacities $q_{0,10rat}$, $q_{0,15rat}$ and $q_{0,20rat}$ for $t_{a2} = 10, 15$ and $20\text{ }^{\circ}\text{C}$, basic cooling capacities $q_{0,10-15}$ for cooling air from $t_{a2} = 15$ to $10\text{ }^{\circ}\text{C}$ and available residual boost part $q_{0,b10-15}$ for cooling air $t_{a2} = 15\text{ }^{\circ}\text{C}$ (a) and available residual boost part $q_{0,b10-20}$ for cooling air $t_{a2} = 20\text{ }^{\circ}\text{C}$ (b).

The values of specific cooling capacities $q_{0,15}$ needed for cooling ambient air to $t_{a2} = 15\text{ }^{\circ}\text{C}$, rational cooling capacities $q_{0,10rat}$, $q_{0,15rat}$ and $q_{0,20rat}$ for cooling ambient air to $t_{a2} = 10, 15$ and $20\text{ }^{\circ}\text{C}$ accordingly, the basic cooling capacity as difference $q_{0,10-15} = q_{0,10} - q_{0,15}$, needed for cooling air from $t_{a2} = 15\text{ }^{\circ}\text{C}$ to $t_{a2} = 10\text{ }^{\circ}\text{C}$, available residual boost cooling capacities $q_{0,b10-15}$ and $q_{0,b10-20}$ are calculated

for climatic conditions in Voznesensk, Nikolaev region in July 2017 (Fig. 2).

As Fig. 2 shows, with cooling the ambient air to $t_{a2} = 15\text{ }^{\circ}\text{C}$ the fluctuations of the current thermal loads $q_{0,15}$ are gradual, that points to significant amount of an excessive cooling capacity in the temperate daily hours. At the same time, when air is cooled from $t_{a2} = 15\text{ }^{\circ}\text{C}$ to $t_{a2} = 10\text{ }^{\circ}\text{C}$, the fluctuations in the thermal load $q_{0,10-15} = q_{0,10} - q_{0,15}$ are comparatively small. Thus, the temperature of cooled air $t_{a2} = 15\text{ }^{\circ}\text{C}$ can be assumed as the threshold temperature for shearing the overall design thermal load on the TIAC system $q_{0,10rat}$ into a comparatively stable ("laminarized") load range $q_{0,10-15}$ and the boost unstable ("turbulent") range of ambient air precooling. So, the stable load value $q_{0,10-15} = q_{0,10} - q_{0,15}$ is chosen as basic stable part $q_{0,10-15} = q_{0,10} - q_{0,15}$ of a design cooling capacity $q_{0,10rat} = 35\text{ kJ/kg}$ (Fig. 1). Accordingly, the remaining part of $q_{0,10rat}$ is used for precooling the ambient air to the threshold temperature $t_{a2} = 15\text{ }^{\circ}\text{C}$ and determined as boost cooling capacity $q_{0,b10-15rat} = 35 - q_{0,10-15}$ (Fig. 2,b). The unstable $q_{0,15}$ thermal load range can be covered by ACh as well as the stable $q_{0,10-15}$ thermal load range – by ECh (Fig.3).

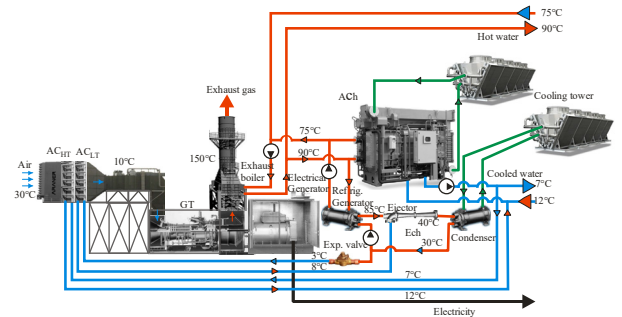


Fig. 3. A hybrid two-stage TIAC system in AECh: AC_{HT} and AC_{LT} – high and low temperature stages of air cooler; Exp. Valve – expansion valve.

As Fig. 2, a shows, the available boost cooling capacity $q_{0,b10-15rat}$ generally covers current thermal loads $q_{0,15}$ for precooling ambient air to $t_{a2} = 15\text{ }^{\circ}\text{C}$. Furthermore, even less available boost cooling capacity $q_{0,b10-20rat}$ also covers the current loads $q_{0,15}$ except quite short periods of daylight hours (Fig.2, b).

The next step in further enhancing the efficiency of TIAC systems and development of their design methodology is issued from the behavior of daily boost thermal load fluctuations. Proceeding from this, a new hypothesis is proposed to divide a design thermal load of unstable boost range $q_{0,b10-15}$ in two parts as $q_{0,b10-15} = q_{0,b10-20} + q_{0,15-20}$ or $q_{0,15rat} = q_{0,20rat} + q_{0,15-20}$ with daily accumulated excess as $q_{0,20exc} = q_{0,b10-20} - q_{0,15}$ of design cooling capacity $q_{0,20rat}$ over cooling needs $q_{0,15}$ for cooling air to $t_{a2} = 15\text{ }^{\circ}\text{C}$ in order to cover daily deficit $q_{0,20def} = q_{0,15} - q_{0,b10-20}$ at peaked loads. On comparing current values of excess $q_{0,20exc}$ and deficit $q_{0,20def}$ as resulting refrigeration energy generated $\Sigma(q_{0,20} \tau)_{exc} = \Sigma(q_{0,b10-20rat} - q_{0,15})\tau$ the hypothesis to reduce a design boost thermal load range $q_{0,b10-15}$ or $q_{0,15rat}$ through using $q_{0,20rat}$ to cover current cooling loads $q_{0,15}$, i.e. practically twice as compared with a design rational value $q_{0,15rat}$ has been approved (Fig.4).

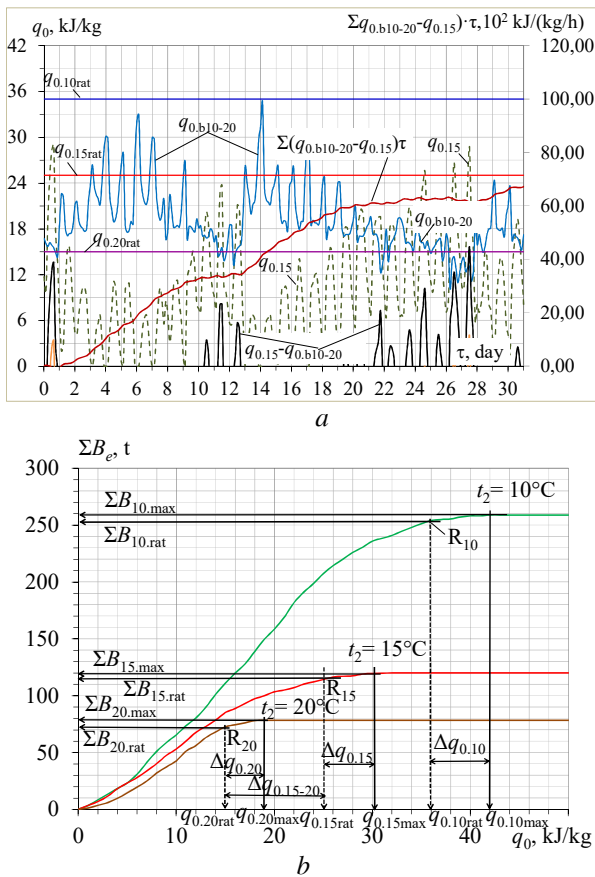


Fig. 4. Current values of thermal load $q_{0.15}$ for cooling ambient air to $t_{a2} = 15^\circ\text{C}$, available specific boost cooling capacity $q_{0.b10-20}$ for cooling air to $t_{a2} = 20^\circ\text{C}$, deficit of design cooling capacity $q_{0.20\text{def}} = q_{0.15} - q_{0.b10-20}$ for cooling air to $t_{a2} = 15^\circ\text{C}$, summarized excess of available refrigeration energy $\Sigma(q_{0.20} \tau)_{\text{exc}} = \Sigma(q_{0.b10-20} - q_{0.15}) \tau$ over required $q_{0.15}$ for $t_{a2} = 15^\circ\text{C}$ (b), annual fuel reduction ΣB_e against specific design cooling capacity q_0 for cooling ambient air to $t_{a2} = 10, 15$ and 20°C : $\Sigma B_{10;15;20}$ – for $t_{a2} = 10; 15$ and 20°C ; $\Delta q_{0.15-20} = q_{0.15\text{rat}} - q_{0.20\text{rat}}$; $\Delta q_{0.10,15,20} = q_{0.10,15,20\text{max}} - q_{0.10,15,20\text{rat}}$; points $R_{10,15,20} - q_{0.10,15,20\text{rat}}$ and $\Sigma B_{10,15,20\text{rat}}$.

There are following correlations used:

reduction of design cooling capacity of ACh boost stage $\Delta q_{0.15-20} = q_{0.15\text{rat}} - q_{0.20\text{rat}}$;

reduction of design cooling capacity of AECh $\Delta q_{0.10} = q_{0.10\text{max}} - q_{0.10\text{rat}}$ and of ACh $\Delta q_{0.15,20} = q_{0.15,20\text{max}} - q_{0.15,20\text{rat}}$;

current deficit of design cooling capacity $q_{0.20\text{def}} = q_{0.15} - q_{0.b10-20}$;

summarized excess of available refrigeration energy $\Sigma(q_{0.20} \tau)_{\text{exc}} = \Sigma(q_{0.b10-20} - q_{0.15}) \tau$.

As Fig. 4, a testifies, the available boost cooling capacity $q_{0.b10-20}$ designed for cooling air to $t_{a2} = 20^\circ\text{C}$ in general case is enough to cover the current cooling demands $q_{0.15}$ for deeper cooling air to $t_{a2} = 15^\circ\text{C}$. The current deficit of design cooling capacity $q_{0.20\text{def}} = q_{0.15} - q_{0.b10-20}$ can be covered through using the daily accumulated excessive refrigeration energy. This statement is also approved by the continuously arising curve of the summarized excess of available design refrigeration energy over its current deficit $\Sigma(q_{0.20} \tau)_{\text{exc}} = \Sigma(q_{0.b10-20} - q_{0.15}) \tau$.

As Fig. 4, b shows, rational designing of TIAC systems provides decrease of installed cooling

capacities of the chillers and TIAC systems in the whole by the values of $\Delta q_{0.10,15,20}$, i.e. by 15 to 20 % compared with their maximum magnitudes $q_{0.10,15,20\text{max}}$, calculated according conventional practice of designing. The rational distribution of the installed cooling capacity of ACh enables to reduce a design boost load by the value $\Delta q_{0.15-20} = q_{0.15\text{rat}} - q_{0.20\text{rat}}$ (Fig. 4, b), i.e. practically twice as compared with $q_{0.15\text{rat}}$.

In temperate climatic conditions the application of rationally designed hybride two-stage TIAC systems with combined AECh enables to provide about 50% higher annual fuel saving $\Sigma B_{10\text{rat}}$ at $q_{0.10\text{rat}}$ as compared with $\Sigma B_{15\text{rat}}$ at $q_{0.10\text{rat}}$ for ACh (Fig. 4,b) and can be considered as a novel prosperous trend in TIAC.

5 Conclusions

A novel trend in TIAC by two-stage air cooling in combined AECh is proposed to provide about 50% higher annual fuel saving in temperate climatic conditions. An advanced methodology is developed to determine a rational design cooling capacities of TIAC systems that provides closed to maximum annual fuel saving and decrease of installed cooling capacity by 15 to 20 % as compared with conventional TIAC designing practice.

A novel approach to designing the TIAC systems through rational distribution of the overall design cooling capacity between unstable ("turbulent") thermal load range for ambient air precooling in the boost high temperature stage of the air cooler by ACh and a stable ("laminarized") load range for further air subcooling to the target temperature in the low temperature stage by ECh. Such two-range distribution of the overall cooling capacity provides the favorable thermal loading conditions for operation of ECh at practically stable loads and realization of the advantages of ACh and ECh in combined AECh (high COP and deep air cooling).

Because of great uncertainty of unstable boost ("turbulent") load range magnitude its design value is determined by a remaining principle as a difference between the overall design cooling capacity for cooling the ambient air to the target temperature and its basic stable ("laminar") load range for subcooling air from a threshold temperature of the ambient air precooled in ACh. Issuing from daily fluctuations of excess and deficit of cooling capacity a new hypothesis to divide a design thermal load of unstable boost range in two load parts with daily accumulated excess of available design cooling capacity to cover daily deficit and to reduce a design boost load practically twice as result.

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